

Tailorable and Repeatable Normal Contact Stiffness via Micropatterned Interfaces

Jack perris¹, Yang Xu^{2*}, Mehmet E. Kartal³, Nikolaj Gadegaard¹ and Daniel M. Mulvihill^{1*}

¹James Watt School of Engineering, University of Glasgow, Glasgow, G12 8QQ, UK

²School of Mechanical Engineering, Hefei University of Technology, Hefei, 230009, China

³ School of Engineering, University of Aberdeen, Aberdeen, AB24 3UE, UK

ABSTRACT

An approach to producing interfaces with tailored and repeatable normal contact stiffness using micropatterned surfaces is developed. A finite element model is first used to design square wave interfaces having a range of stiffnesses and these are fabricated in polycarbonate via a microfabrication process. Results demonstrate that the contact stiffnesses of the fabricated interfaces are both tailorable and repeatable. The approach can be broadened to other materials and is useful for applications requiring specified interface stiffness. Finally, even with these deterministic interfaces, we show that low levels of roughness on the surface features is sufficient to produce a load-dependent contact stiffness at lower loads. Therefore, tailorability is mostly applicable above this limit where total contact stiffness converges to a load-independent value.

Keywords: contact stiffness, structured surfaces, micropattern, microfabrication.

Corresponding authors: *Daniel M. Mulvihill (daniel.mulvihill@glasgow.ac.uk) and Yang Xu (Yang.Xu@hfut.edu.cn)

1. Introduction

Normal contact stiffness K is defined as the instantaneous rate of change of normal load P with the relative approach u of two surfaces in contact (i.e. $K = |dP/du|$). Depending on how it is measured, it may be calculated based on deformation caused purely by the rough surface topography or it can be based on a combination of rough surface deformation and some amount of bulk material deformation occurring near the interface. The stiffness of a contact interface can be important in the design of mechanical systems depending on the application. It can affect the stability of a bone-implant interface [1], the vibration behaviour of joints [2], the electrical and thermal conductance of a contact interface [3, 4], the behaviour of robotic gripping systems [5, 6], and the performance and control of precision machinery [7]. Various advances have been made in the measurement of contact stiffness and a number of approaches are available. Successful methods have been based on: ultrasound [8-10], laser interferometry [11], contact resonance [12] and digital image correlation [13-15]. Two main approaches have been taken to analytically modelling rough surface contact stiffness: asperity-based models [16, 17] built-up from the contact mechanics of single asperity interaction (i.e. Greenwood and Williamson's framework [18]) and the mean field approach based on overall surface statistics (i.e. Persson's approach [19]). Several advanced (and computationally intensive) computational approaches have also been deployed (e.g. finite element method [20], boundary element method [21] and Green's function molecular dynamics [22]). The models and experiments generally exhibit either a linear or power law contact stiffness variation with normal load depending on the contact details [22]. However, the accuracy of contact stiffness prediction for *real* engineering surfaces remains challenging due to the complexity of real contact scenarios and the simplifications inherent in the various modelling approaches. Repeatability is also a problem for real contact interfaces – e.g. a joint giving a different contact stiffness when reassembled. This makes it difficult to design interfaces where contact stiffness is a critical parameter. One example is the modelling of aeroengine vibrations. Very accurate finite element models exist to describe bulk component behaviour, but key interface parameters such as contact stiffness have to be empirically determined [23]. The random multiscale nature of the roughness present on most typical engineering surfaces is what makes the problem difficult. For example, Barber [24] showed that statistical fluctuations make a prediction of stiffness difficult if there is no separation between the scales of the macroscopic object and the longest wavelength of the roughness.

In this paper we explore the possibility of using micropatterning to generate more deterministic surfaces with tailored and repeatable contact stiffness. The idea of using structured interfaces to tailor interface properties was recently explored for friction in Bin Jaber et al. [25] and for adhesive joints in Hamilton et al. [26]. The most relevant paper in relation to contact stiffness is that by Li *et al.* [27]. Li *et al.* explored the contact stiffness of an array of rigid cylindrical micropillars impinging a flat deformable surface. Using elastic flat-punch theory, they showed how interaction between impinging pillars leads to a contact stiffness dependent on, not just pillar spacing, but also on the number of contacting pillars (or equivalently, the apparent size of the overall nominal contact region). In this paper, we begin with a finite element approach to first design a set of micropatterned surfaces with varying instances of contact stiffness. We then use photolithography, etching and injection moulding steps to fabricate these surfaces (in polycarbonate - PC) and explore the extent to which the contact stiffnesses of the real *as-produced* surfaces replicate the design surfaces. Digital image correlation local to the contact interface is used to accurately measure displacements (and, hence, contact stiffness). We also explore some of the key factors effecting the realisation of an adjustable tailored contact stiffness: apparent contact area (or equivalently feature width in this case), feature geometric imperfections and the presence of nano roughness on the feature tops of the micropatterned surfaces. The repeatability of the contact stiffness for different interface sample sets produced using the same mould tooling is also investigated.

2 Design interfaces

A square wave pattern in contact with a flat surface of the same material (Fig. 1) was used for the investigation – mainly because the square wave topography minimises some of the complexity that can arise in fabrication and makes adjustment of surface feature parameters relatively straightforward. A thermoplastic polymer (Polycarbonate) was used for both interface samples as highly accurate patterns can be injection moulded (see Section 3). The unstructured flat-on-flat case was also included as a benchmark. We first defined a series of square wave instances designed to produce a range of different *design* normal contact stiffnesses. This was done by adjusting the square wave feature width λ_f while holding the period ($\lambda_p = 500 \mu m$) and overall nominal contact size ($10 \times 10 \text{ mm}^2$) constant. The feature

width was adjusted to give nominal contact area ratios of 0.2, 0.4, 0.6, 0.8 and 1 as shown in Fig. 1. Contact area ratio is defined here as the total nominal feature contact area A_{nf} (i.e. the tops of the features) divided by the nominal area A_n (i.e. total enclosed planar area of $10 \times 10 \text{ mm}^2$). Note A_{nf}/A_n is equivalent to λ_f/λ_p . Feature height was held constant at $h = 50 \text{ }\mu\text{m}$. Design dimensions are given in Table 1. A linear elastic finite element model (Fig. 2) was then used to determine the associated design contact stiffness for each case and these are listed in Table 2. For calculation of the FE contact stiffness, node pairs on either side of the interface separated by the same distance $d = 200 \text{ }\mu\text{m}$ as for the calculation of experimental contact stiffness (described later in Section 4 and Fig. 6) were used. This measure of contact stiffness, of course, includes a bulk material contribution. As shown (In Table 2), the various surface instances give contact stiffnesses ranging from 120.4 to 238.2 kN/mm. The challenge is then to determine the extent to which it is possible to replicate these stiffness values repeatably in fabricated samples.

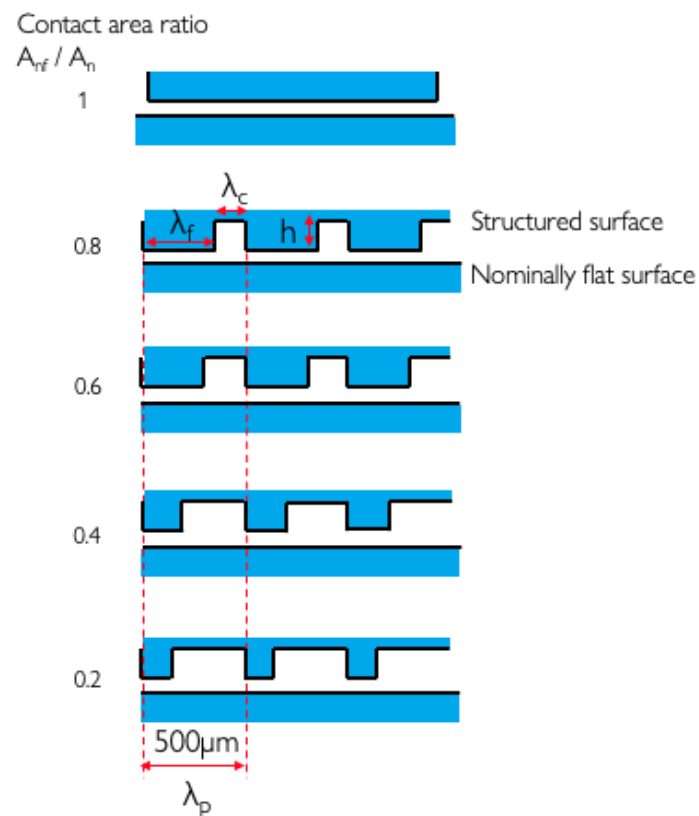


Fig. 1: Schematic representation of square wave instances at different design contact area ratios. Contact area ratio

$$A_{nf}/A_n = \lambda_f/\lambda_p. \text{ Overall nominal contact area is } 10 \times 10 \text{ mm}.$$

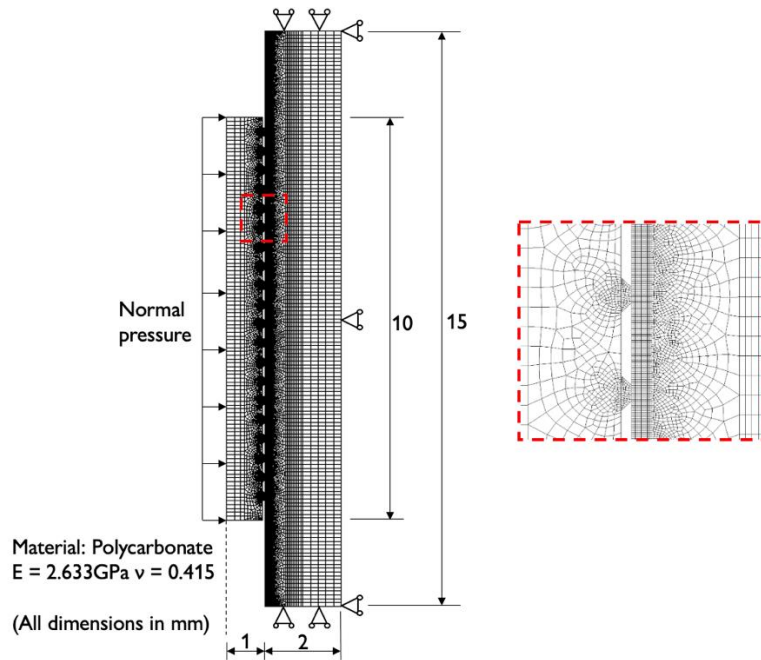


Fig. 2: Finite element model (linear elastic) for determination of contact stiffnesses.

3. Fabrication

Both interface samples (each polycarbonate) were fabricated via injection moulding. This section outlines the steps involved. The square wave pattern was first produced on a silicon master. Standard photolithography was used to generate a grid pattern on a photoresist coating. Deep reactive ion etching (DRIE) was then used to etch the exposed regions in a highly directional process resulting in highly vertical sidewalls. The process is outlined in Fig. 3.

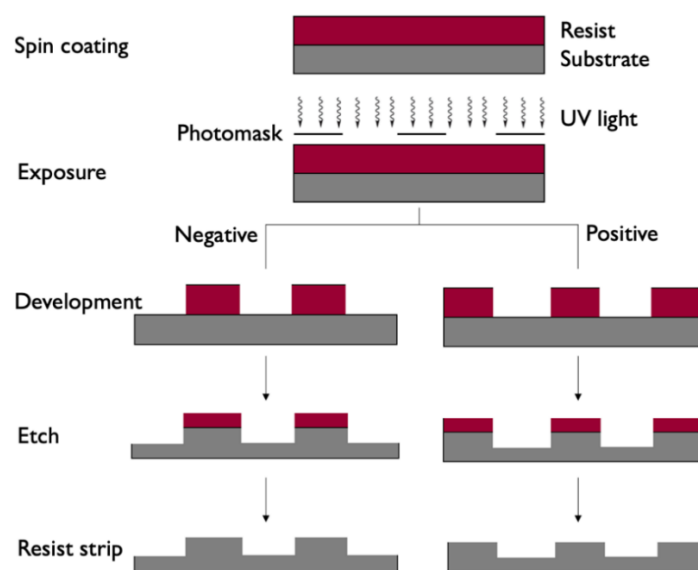


Fig. 3: Overview of silicon square wave microfabrication: development of a pattern in a photoresist coating via photolithography followed by etching of exposed regions to produce the square wave pattern in the silicon itself.

The silicon master is not suitable for use as a mould inlay. Therefore, the second step in the process involves transferring the pattern from the silicon master into a material with suitable thermal properties and durability for use as a mould inlay. This was done using an adapted nanofabrication protocol: Fig. 4 conveys the key stages. The silicon master is covered in an anti-stick layer (to enable separation of the wafer and stamp material) with a substance known as the working stamp material spun on top of the wafer (Fig.4a). The wafer is then placed in a tool where the working stamp layer is imprinted onto a new transparent layer of Polyethylene terephthalate (PET) and subsequently UV-cured (Fig.4b), producing the inverse pattern from the silicon wafer on the new PET sheet (Fig. 4c). The structured foil is then laser cut to the size required for the mould inlay.

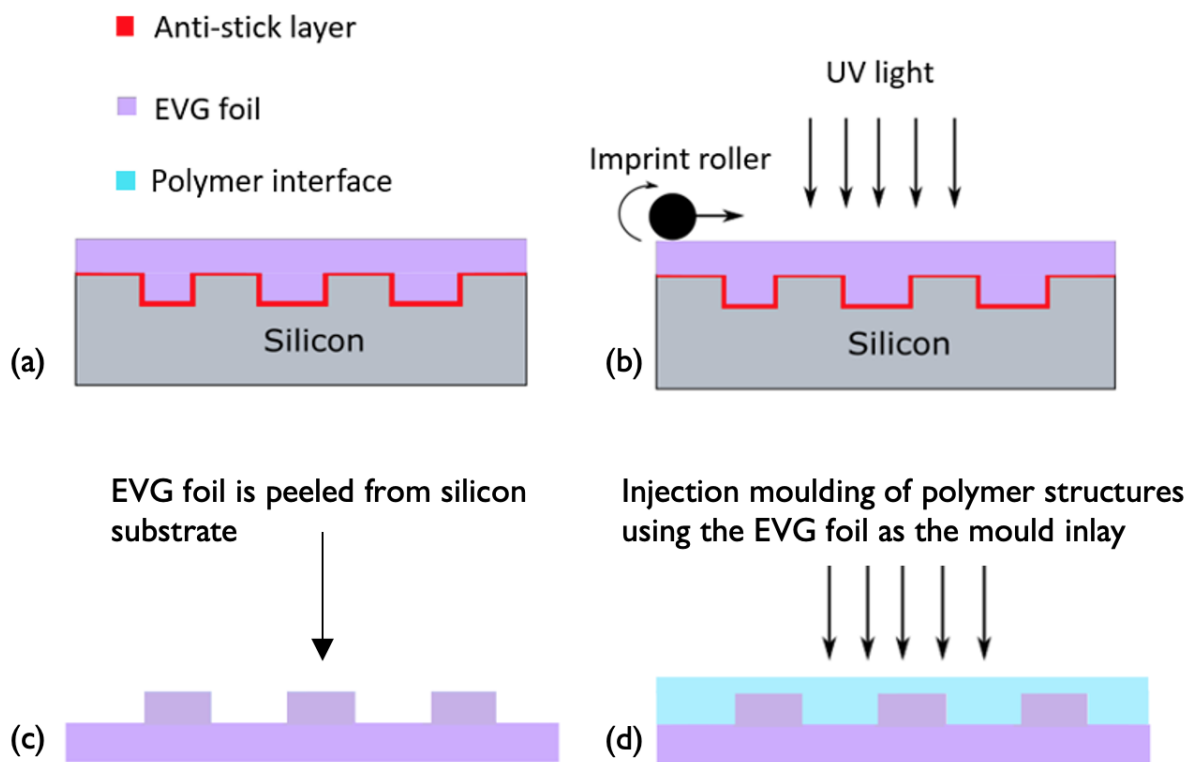


Fig. 4: Creation of mould inlay and injection moulding of polycarbonate samples: (a) spinning of a working stamp EVG foil material (purple) on top of the structured silicon master – an anti-stick layer (red) is spun initially to enable separation from the silicon substrate; (b) imprinting of the micro-features from the silicon to the EVG foil followed by UV curing to solidify the mould insert; (c) imprinted micropatterned EVG foil and (d) injection moulding of polycarbonate samples (blue) using the mould insert from (c).

The final step in the fabrication process uses injection moulding to obtain the micro-structured pattern (Fig. 4d) in polycarbonate. The key parameters that characterise the moulding process are mould temperature, polymer injection temperature, injection velocity,

pressure and the cooling time. Optimising the moulding conditions ensures sufficient filling of the micro-cavities to replicate the geometric fidelity from silicon master to polycarbonate. Injection moulding was performed using an Engel Victory 28 fully hydraulic injection moulding machine with a melt temperature of 270°C and a tool temperature of 70°C. The PET foil supporting the pattern was laser cut to dimensions of 24.5 mm x 24.5 mm and placed inside a tool hardened steel frame. The polycarbonate was dried for a minimum of 2 hours at 110°C in a vacuum oven prior to moulding. More extensive information on the moulding route used here is given in Hamilton et al. [28].

The samples were characterised during the fabrication process to allow analysis and comparison of the stages in the manufacturing process. The feature widths, heights, and channel widths were all measured using optical profilometry (Contour GT, Bruker, US). The critical measurements were taken from the silicon masters, moulding inlays, and polycarbonate replicas. Scans were taken at five locations across the respective samples. Multiple measurements of the critical dimensions were then taken from each scan of the samples.

4. Experiments

A micromechanical test machine (5 kN Dual leadscrew, Deben UK) was used to apply normal load to the samples via custom designed self-aligning fixtures (Fig. 5). A nominally flat polycarbonate sample was glued to the flat platen on the left side of the arrangement, and the polycarbonate countersurface (patterned or flat) was glued to the platen on the right side attached to a ball and socket joint. The ball and socket joint facilitates self-alignment of the interface to ensure uniform contact. The fixtures were also designed so that the central (perpendicular) axis of the polymer samples is in-line with the machine's load cell line-of-action. Tests were conducted with ambient conditions of temperature, pressure and humidity and samples were cleaned via IPA and compressed air prior to testing. An initial load of 5 N is used to align the surfaces. The contact is then loaded up to 1500 N at a rate of 0.5 mm/min (i.e. quasi static) and the load was recorded via the machine's load cell. Relative normal interface displacement was measured local to the interface using digital image correlation (DIC) by tracking the relative approach of target pairs close to the interface (separated initially by approximately $d = 200 \mu\text{m}$) as illustrated in Fig. 6. Five corresponding target pairs were used to facilitate calculation of an average value (only 3 are shown in Fig. 6). To capture the

images for DIC, a side-on video of the interface region was captured using a digital camera (PL-D732 2.2MP, Pixelink) and high magnification adjustable lens system (Zoom 6000, Navitar USA) positioned above the micro-tester. The field of view was 6.3 x 4.7 mm with 640 x 480 pixels giving an image resolution of 9.8 μm . The lens settings selected allowed for sufficient resolution while affording a large enough depth of field to keep the image in focus during the test. Imetrum Video Gauge DIC software was then used to calculate relative displacements from the videos. The software is capable of detecting displacements of $1/200^{\text{th}}$ of a pixel so the minimum resolvable displacement will be about 0.05 μm . The same sampling rate (0.5s) was used for both load measurement and image capture allowing ease of synchronisation of load and displacement. Tests on two separate pairs of samples were each repeated five times for each of the contact area ratio cases of $\lambda_f/\lambda_p = 0.2, 0.4, 0.6, 0.8$ and 1.

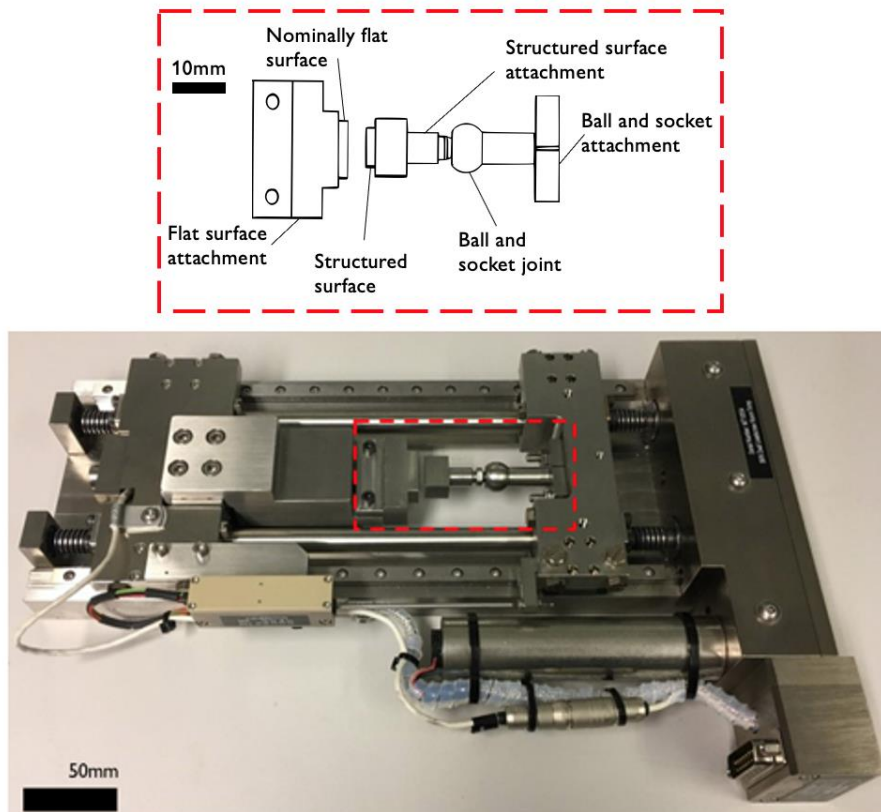


Fig. 5: Experimental test rig for application of normal load: micromechanical test machine with custom designed self-aligning fixtures (dashed red box).

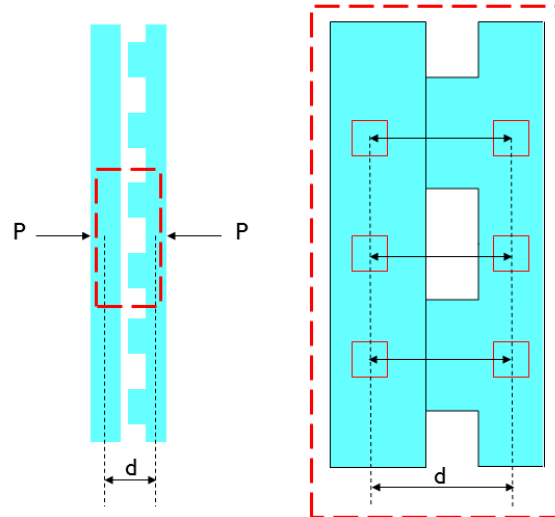


Fig. 6: Schematic representation of specimens prior to contact and zoomed-in region illustrating digital image correlation (DIC) target pairs for tracking relative displacements. The relative displacement of the interface (during the experiment) is the change in d (i.e. Δd). The value of d is initially approximately $200\ \mu\text{m}$ ($100\ \mu\text{m}$ on each side). Five target pairs were used with only three shown here.

5. Results and discussion

Although we started with ‘design’ values for the dimensions of the structured surface pattern, the ‘as-produced’ surfaces in polycarbonate differ considerably from these. Table 1 reports the changes in key surface dimensions (having a bearing on contact stiffness) from design values to as-produced polycarbonate (PC). Dimensions are given for feature depth, feature width and contact area ratio. For feature depth and width, the intermediate silicon master value is also indicated. Going from design to as-produced PC, feature depth increases by a maximum of up to 7.9% (0.6 design contact ratio case), whereas contact area ratio reduces by up to 20% (0.2 design contact ratio case). The reason for this is mostly related to elongation in feature depth and shrinkage at the tops of the features during the injection moulding stage. This is due to a combination of feature stretching during part ejection and polymer shrinkage during cooling. Note, from here on, we thus refer to two contact area ratios: design and as-produced. There will be some dimensional change in realising the intermediate master surfaces in silicon, but most of the change occurs during the polycarbonate moulding step (see Table 1). To account for this change in dimensions, the FE analysis was carried out *both* for the original design profile and for a modified version based on ‘as-produced’ dimensions. Fig. 7 depicts this distinction. Fig. 7a indicates the design waveform and Fig. 7b describes the modified waveform. The modified waveform is based on

as-produced dimensions and includes linear sloped edges to approximate the non-uniform feature shrinkage. An SEM image of a typical cleaved cross-section of an as-produced PC sample is also shown in Fig. 7c for comparison. In the remainder of the paper, a distinction is made between the FE model using the ‘as-designed’ surface profile and the one using the approximation of the ‘as-produced’ surface profile. Note that, although there are some significant differences between the design and as-produced dimensions here, it would actually be straightforward to adjust and refine the fabrication process to produce dimensions far closer to the desired dimensions.

Table 1: Changes in key structured surface dimensions (with a bearing on contact stiffness) from design values to as-produced polycarbonate (PC) surfaces. Table shows evolution of feature depth h , feature width λ_f and contact area ratio. For feature depth and feature width, the corresponding dimension for the intermediate silicon master is also indicated.

Design (μm)	Feature depth (h)			Design (μm)	Feature width (λ_f)			Design	Contact area ratio	
	Silicon (μm)	as-produced PC (μm)	% Increase		Silicon (μm)	as-produced PC (μm)	% Decrease		as-produced PC	% Decrease
50	50.8 ± 0.3	52.3 ± 0.7	4.5	120	126.8 ± 2.9	80.6 ± 1.9	32.8	0.20	0.16	20
50	52.5 ± 0.3	52.8 ± 0.6	5.7	225	225.3 ± 2.6	171.4 ± 3.2	23.8	0.40	0.34	15
50	52.8 ± 0.4	53.9 ± 0.7	7.9	330	323.8 ± 2.3	275.4 ± 3.8	16.6	0.60	0.55	8.3
50	49.7 ± 0.4	51.1 ± 0.5	2.2	440	439.9 ± 2.6	387.8 ± 3.7	11.9	0.80	0.77	3.8

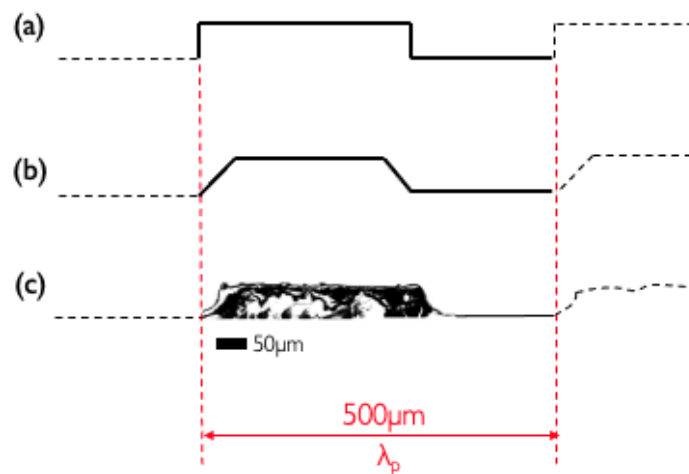


Fig. 7: (a) as designed surface waveform based on design dimensions, (b) modified surface waveform with graduated edges based on ‘as-produced’ dimensions, and (c) SEM image of a cleaved cross-section of an ‘as-produced’ structured polymer sample (with scale bar included).

Fig. 8a shows a representative plot of normal load versus relative displacement from the experiments (case shown here for design contact area ratio of 0.2). Plots for each of the five target pairs are shown together with the average trace. Note that there is some variation in the total amount of relative displacement that occurs at each target pair which may be dependent on small discrepancies in the heights of the features or with the interface alignment (or indeed on whether a target pair lines up with a square wave feature or a gap). However, the traces appear to be parallel indicating similar contact stiffness response. Fig. 8b gives the corresponding plot of normal contact stiffness versus normal load (as calculated from the derivative of the average trace in Fig. 8a). Note that the small-scale oscillations apparent in Fig. 8a had to be smoothed out before taking the derivative. Three zones are apparent in Fig. 8a: (i) initial large displacement due to bending to accommodate waviness and lack of flatness, (ii) load dependent increasing interface stiffness due to flattening of surface topography (the curved part) and (iii) saturated interface stiffness (the linear part). Fig. 8b clearly shows the increasing contact stiffness followed by a plateau. The dashed line in Fig. 8b represents the slope of the linear (saturated stiffness) region in Fig. 8a. It is reasonable to assume that the saturation in the contact stiffness coincides with the effect of the surface roughness (on the tops of the features) being sufficiently flattened such that the bulk material deformation between the measured points dominates.

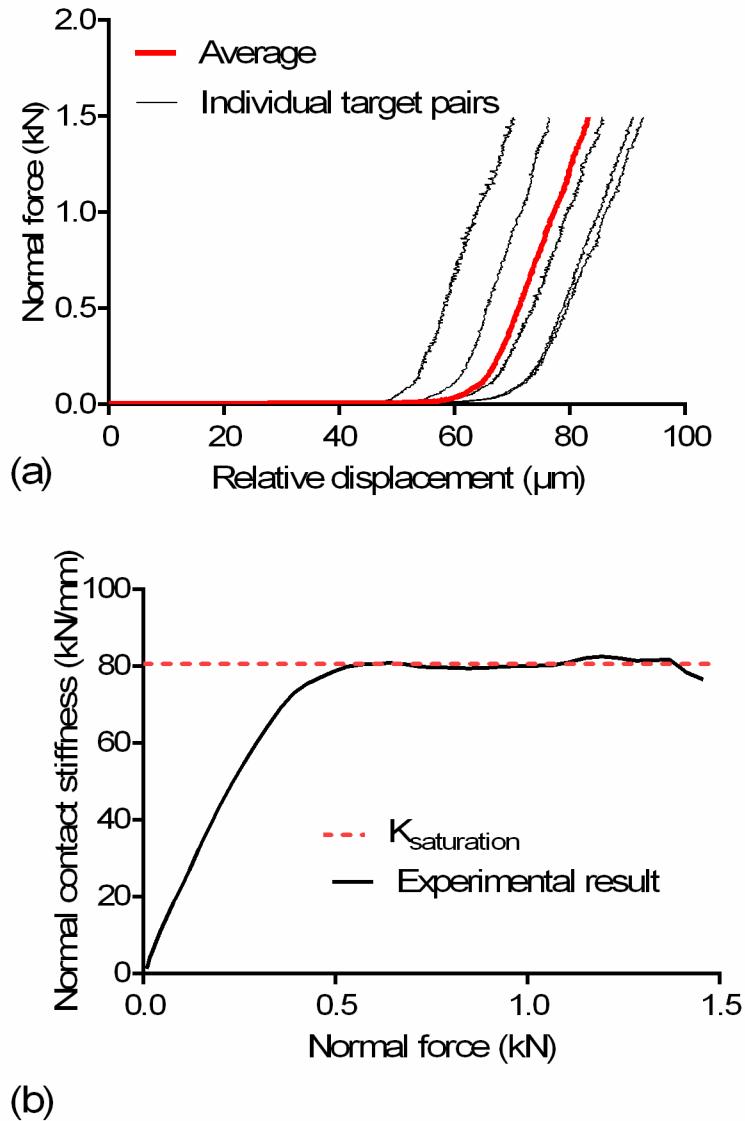


Fig. 8: (a) Normal force versus relative normal displacement from a typical test showing the result at each of the five target pairs as well as the average trace (contact stiffness at saturation is determined as the slope of the linear region) and (b) Normal contact stiffness versus normal force for the same test as determined from the derivative of the average trace (dashed line is the slope of the linear region in (a)). Design contact ratio is 0.2 (i.e. as produced contact ratio of 0.16).

Fig. 9 compares normal contact stiffness versus load for each of the five design contact area ratios (0.2, 0.4, 0.6, 0.8 and 1 corresponding to measured values of 0.16, 0.34, 0.55, 0.77 and 1). Despite some variation, in each case, the curves plateau to a saturated contact stiffness corresponding to the slope of the linear region of the corresponding average trace of load versus displacement (this value is indicated by the dashed red line).

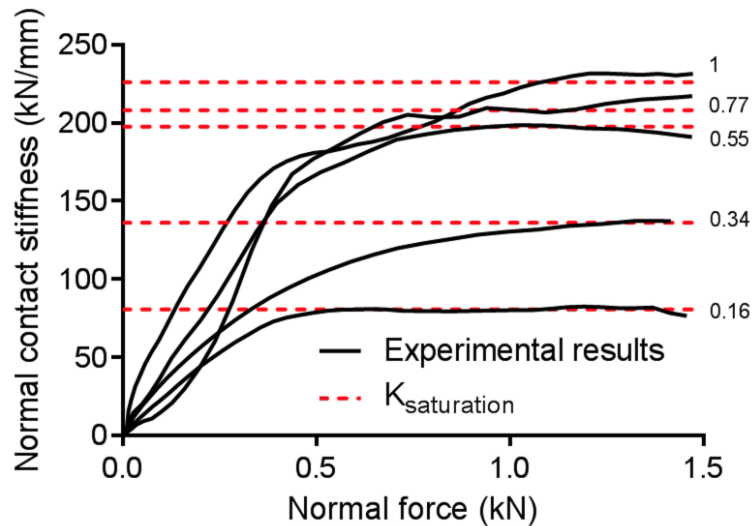


Fig. 9: Normal contact stiffness versus load for each of the five contact area ratio cases. The curves are determined from the derivative of the corresponding average force-displacement trace and the dashed lines represent the slope of the linear part of the trace. Note: the design contact area ratios 0.2, 0.4, 0.6, 0.8 and 1 correspond to the measured values of 0.16, 0.34, 0.55, 0.77 and 1 shown on the plot.

It is now reasonable to compare the saturated experimentally measured stiffness values to the stiffness values from the FE models (as the surface topography effect present on the tops of the features in the experimental results should be minimised at saturation). The comparison is indicated in Fig. 10 and Table 2 where the measured results are compared with both ‘as designed’ and modified ‘as produced’ FE results. Note, the experimental contact stiffness results in Fig. 10 and Table 2 are the mean values of the slopes of the linear region of the load-displacement traces of the five target pairs. Owing mostly to the dimensional changes introduced during fabrication of the structured surfaces (see Table 1), the experimental contact stiffness values differ somewhat from the ‘as designed’ FE values: they are 5.1 to 35.1% lower according to Table 2. However, when the dimensional changes are accounted for (approximately) in the FE model (the ‘as produced’ case), this discrepancy reduces to a 5 to 12.9% drop. Indeed, Fig. 10 shows that the experimental stiffness values closely follow the ‘as produced’ FE results. In fact, the larger deviation between the measured results and the ‘as designed’ case is not really a problem as the fabrication process can easily be adjusted and refined to give surface dimensions matching the design case far more closely. In that sense, the agreement of the measured results with the ‘as produced’ FE results are actually a good indication of the feasibility of tailoring contact stiffness. Crucially, the measured stiffness values closely follow the FE designs in terms of both trend and magnitudes

(Fig. 10). The premise is rather simple: as the contact area ratio is reduced, the width of the features is reduced, there is less material to resist the normal load and the stiffness of the square wave interface in the normal direction is reduced. As the contact area ratio was reduced from unity to 0.2, it was possible to reduce the contact stiffness by almost three times (Table 2). It is also very promising that the experimental results are very repeatable. Tests for each contact area ratio were repeated five times on each of two separate pairs of samples (i.e. 10 tests at each contact area ratio) and the error bars in Fig. 10 and standard deviations in Table 2 indicate excellent repeatability.

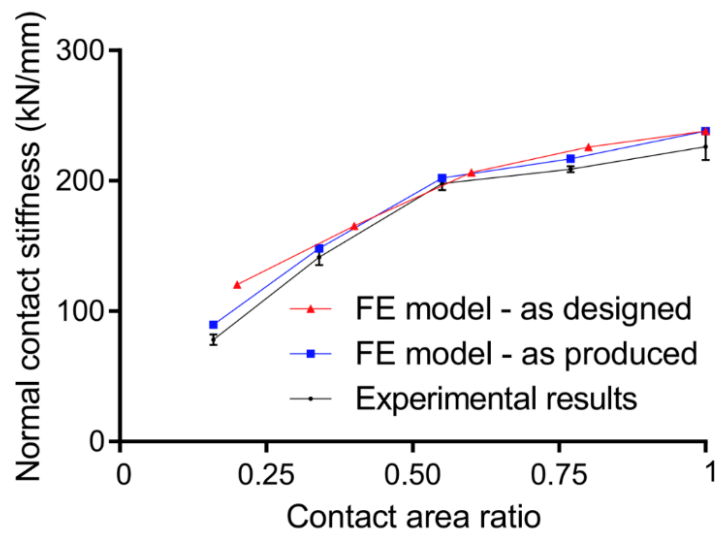


Fig. 10: Measured (total) normal contact stiffness (at saturation) versus nominal contact area ratio with comparison to ‘as designed’ and modified ‘as produced’ FE representations (Note: Contact area ratio = $\frac{\lambda_f}{\lambda_p}$)

Table 2: Comparison of experimentally measured (total) normal contact stiffness (at saturation) with finite element determined contact stiffness for ‘as-designed’ and modified ‘as-produced’ cases.

FE – as designed		FE – as produced		Experimental			
Contact area ratio	Normal contact stiffness, K (kN/mm)	Contact area ratio	Normal contact stiffness, K (kN/mm)	Contact area ratio	Normal contact stiffness, K (kN/mm)	% drop from ‘as-designed’ FE’	% Drop from ‘as-produced’ FE
0.2	120.4	0.16	89.7	0.16	78.1 ± 4.0	35.1	12.9
0.4	165.4	0.34	148.2	0.34	141.5 ± 5.9	14.5	4.5
0.6	206.4	0.55	202.2	0.55	197.9 ± 5.1	4.1	2.1
0.8	225.9	0.77	216.8	0.77	208.9 ± 2.2	7.6	3.6
1	238.2	1	238.2	1	226.2 ± 10.3	5.1	5

The possibility to practically fabricate interfaces with tailorable and repeatable contact stiffness (as demonstrated here for polycarbonate surfaces) has wide implications. There are several engineering scenarios where accurate control of contact stiffness is required – as mentioned, the stability of a bone-implant interface, the vibration behaviour of joints, the electrical and thermal conductance of a contact interface, the behaviour of robotic gripping systems, and the performance and control of precision machinery. The approach here can easily be expanded to fabrication of structured surfaces in other polymers via injection moulding or 3D printing and indeed in metals via techniques such as metal 3D printing or electroplating.

To better explain the transition of normal contact stiffness versus normal load from load dependent to a near constant value (i.e. as in Figs. 8 and 9), the total normal contact stiffness K can be decomposed as follows using the approach taken by Medina *et al.* [29]:

$$\frac{1}{K} = \frac{1}{K_{\text{saturation}}} + \frac{1}{K_{\text{roughness}}} \quad (1)$$

where $K_{\text{saturation}}$ is equivalent to the bulk stiffness when the roughness is completely flattened (i.e. at sufficiently high normal load), and $K_{\text{roughness}}$ is the contribution from roughness alone. The roughness in our experiments is the roughness inevitably present on the tops of the square wave features. In the scope of linear elastic contact, the finite element model shows that $K_{\text{saturation}}$ is nearly constant regardless of the magnitude of the normal load. Therefore, $K_{\text{roughness}}$ must monotonically increase with the normal force. The monotonic increase of $K_{\text{roughness}}$ can be confirmed (for the surfaces used in the tests here) using a linear elastic boundary element method (BEM) model developed by Polonsky and Keer [30]. The rough surface topography of the polycarbonate samples was measured using an optical profilometer (Contour GT, Bruker, US) at seven different locations each with sample area of $133.5 \times 176.7 \mu\text{m}^2$ and in-plane resolution of $0.128 \mu\text{m}$. The corresponding rough surface statistics are given in Table 3. R_a and R_q are center-line average and root mean square roughness; σ_s is the standard deviation of summit heights of all asperities and κ_s is the mean summit curvature of all asperities.

Table 3: Statistics of the measured rough surface: R_a and R_q are center-line average and root mean square roughness, σ_s is the standard deviation of summit heights of all asperities and κ_s is the mean summit curvature of all asperities.

R_a (μm)	R_q (μm)	σ_s (μm)	κ_s ($1/\mu m$)
0.026 ± 0.007	0.048 ± 0.008	0.040 ± 0.004	0.071 ± 0.009

Assuming the mating surface (polycarbonate samples) share the same roughness topography, the average interfacial gap and percentage of real contact area associated with a sequence of increasing normal loads (up to 700N) are calculated in Fig. 11a. The average interfacial gap here is the mean gap between two deforming (rough) contact surfaces (i.e. the rough surfaces representative of the tops of the features). The (mean) average interfacial gap gradually reduces with normal force. $K_{roughness}$ can then be determined as the absolute derivative of normal force with respect to the average interfacial gap which can be represented by the reciprocal of the slope of the average interfacial gap to normal load curve in Fig. 11a. The interfacial stiffness exponentially increases up to infinity once the roughness is completely flattened where the average interfacial gap becomes vanishingly small. However, this exponential increase is slowed down by plasticity and the multi-scale nature of roughness. Therefore, even with these deterministic structured surfaces, low levels of surface roughness present on the tops of the features ($R_a \approx 0.026 \mu m$) is responsible for the load dependent contact stiffness in the lower load regime. This means that the tailorable aspect discussed above is mostly only applicable above the load required to produce a load independent saturated contact stiffness. Now, using Eq. 1, the total normal contact stiffness K predicted by the BEM can now be obtained using the $K_{roughness}$ predicted by the BEM (at each load) and using $K_{saturation} = 80.5 \text{ kN/mm}$ (from the experiments). Compared with the experimental results shown in Fig. 11b, the BEM prediction shows a similar convergent trend as normal force increases, but with a much steeper slope. This is a strong signal that $K_{roughness}$ is overestimated by the BEM. The major reason for this is probably the neglect of the plastic deformation at asperity level. The plastic deformation of the rough (isotropic) surface can be quantified using the plasticity index [18]:

$$\psi = \frac{E^*}{H} \sqrt{\sigma_s \kappa_s}$$

where $E^* = \frac{E_{PC}}{2(1-\nu_{PC}^2)}$ is the effective modulus of PC contact pair; H is the hardness of PC which can be estimated using $H = 2.8 S_y$ where S_y is the yield strength of PC. Using $E_{PC} = 2350$ MPa, $S_y = 60$ MPa and values of σ_s and κ_s in Table 3, we get the plasticity index $\psi = 1.41 > 1$ which indicates that the asperity deformation is predominantly plastic. According to Fig. 11b, the experimental normal contact stiffness starts to converge at the normal force of 400 N where the real contact area is predicted by the BEM to only occupy approximately 13% (Fig. 11a) of the entire nominal contact area. If plastic deformation is included, a much higher percentage of real contact area may be achieved, as well as a lower interfacial gap. This would have the effect of decreasing the interfacial stiffness and softening the total normal BEM contact stiffness curve towards the experimental curve.

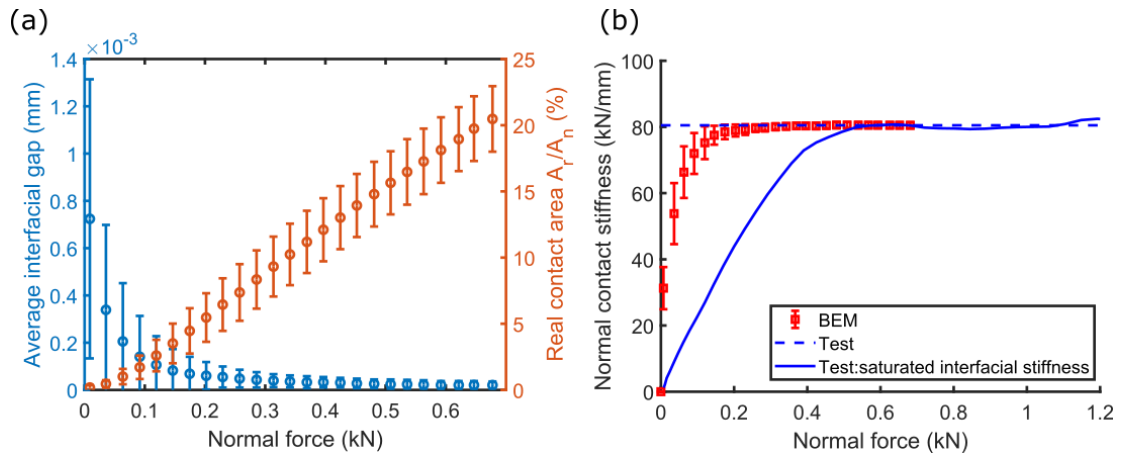


Fig. 11: (a) Evolution of average interfacial gap (left y-axis) and percentage of real contact area (right y-axis) with respect to the normal force for the BEM model of the rough surface polycarbonate contact pair and (b) Comparison of the measured and numerical (BEM) total normal contact stiffness results. Note: The average interfacial gap here is the mean gap between two deforming (rough) contact surfaces (i.e. the rough surfaces representative of the tops of the features).

6. Conclusions

A novel practical approach to realising interfaces with tailored and repeatable normal contact stiffness using micropatterned surfaces has been developed. The paper uses a finite element model to first design square wave interfaces having a range of controllable contact stiffnesses and then fabricates these in polycarbonate via a microfabrication process involving steps such as photolithography, etching, micro-imprinting and injection moulding. The ‘as produced’ surface dimensions, and consequently the measured contact stiffnesses, differed somewhat from the original starting design values, but this is not a problem because the fabrication

process can easily be refined to produce surface profiles having the desired dimensions. In this case, rather than adjusting the process, we adjusted the FE model to accommodate changes introduced during fabrication. Results indicate that the experimental contact stiffness values closely follow the FE predictions in terms of magnitudes and trend. The results were also highly repeatable both between new test samples and between repeat tests. The idea is simple: by adjusting the widths of the features, the contact stiffness can be controlled. In the present work, it was possible to reduce the contact stiffness by almost three times going from a contact area ratio of unity to 0.2. This capability to produce interfaces with tailorable and repeatable contact stiffness will be useful in several engineering applications where contact stiffness is important: bone-implant interfaces, vibration of mechanical joints, operation of robotic gripping systems and the performance and control of precision devices. The approach presented here can be expanded to other polymers via injection moulding and 3D printing and to metals via approaches such as metal 3D printing.

Even with these deterministic micro-structured surfaces, surface roughness present on the tops of the features ($R_a \approx 0.03 \mu\text{m}$) was shown to produce a load dependent contact stiffness in the lower load regime. In the tests in the present work, the higher loads applied were sufficient to flatten the roughness and produce a load independent saturated contact stiffness. Therefore, the tailorable aspect is mostly applicable beyond a certain load limit when the stiffness reaches its plateau value. Comparison of the experimental results with elastic predictions made by applying the boundary element method to the measured roughness shows that the elastic prediction produces a significantly more rapid increase in total contact stiffness with load than the experimental measurements. Therefore, it is suggested that plastic deformation of the rough surface topography in the experimental case is responsible for the more compliant measured interface response.

Declarations

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Conflicts of interest/Competing interests (There are no conflicts of interest/competing interests)

Availability of data and material (Data will be made available via an online University data repository)

Code availability (not applicable)

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